Arrays of flow channels with heat transfer embedded in conducting walls

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Abstract

Here we illustrate the free search for the optimal geometry of flow channel cross-sections that meet two objectives simultaneously: reduced resistances to heat transfer and fluid flow. The element cross section and the wall material are fixed, while the shape of the fluid flow opening, or the wetted perimeter is free to vary. Two element cross sections are considered, square and equilateral triangular. We find that the two objectives are best met when the solid wall thickness is uniform, i.e., when the wetted perimeters are square and triangular, respectively. We also consider arrays of square elements and triangular elements, on the basis of equal mass flow rate per unit of array cross sectional area. The conclusion is that the array of triangular elements meets the two objectives better than the array of square elements.

Keywords: Constructal design, heat exchanger, array of channels, flow channel, square cross section, triangular cross section, multiple objectives, morphing.

Nomenclature

$A_{s,f}$	cross sectional area, m^2
c_p	specific heat at constant pressure, $J\ kg^{-1}\ K^{-1}$
D_h	hydraulic diameter, m
f	friction factor
$\bar{h}$	effective heat transfer coefficient, $W\ m^{-2}\ K^{-1}$
$k_{s,f}$	thermal conductivity, $W\ m^{-1}\ K^{-1}$
$\tilde{k}$	thermal conductivity ratio $\tilde{k} = k_s / k_f$
L_c	distance between centers, Fig. 1c, m
$L_{s,t}$	side length, m
L_x	length of channel, m
$\dot{m}$	mass flow rate, $kg\ s^{-1}$
$\dot{m}''$	mass flow rate per unit area of array cross section, $kg\ s^{-1}\ m^{-2}$
N	number of heat transfer units
Nu	Nusselt number
p	wetted perimeter, m
Po	Poiseuille constant
Pr	Prandtl number
R	radius of rounded corner, Fig. 1c, m
Re_{D_h}	Reynolds number
R_f	dimensionless fluid flow resistance
R_{th}	dimensionless thermal resistance
T_w	external walls temperature, K

T_{in} inlet temperature, K

T_{out} outlet temperature, K

U average fluid velocity, $m\ s^{-1}$

Greek symbols

ϕ dimensionless flow area, Eq. (6)

ψ dimensionless perimeter, Eq. (6)

ν kinematic viscosity, $m^2\ s^{-1}$

ρ fluid density, $kg\ m^{-3}$

Subscripts

f fluid

s solid

s square element

t triangular element

1. Introduction

The shape of the flow channel cross-section plays an important role in the flow and heat transfer performance of heat exchangers. The main purpose of optimizing the shape of the flow channel is to simultaneously enhance the heat transfer rate and to reduce the pumping power requirement. Recent studies have focused on optimizing the flow architecture in order to enhance the heat transfer rate [1-6]. Several studies have demonstrated the effect of the wetted perimeter shape on the flow performance [7-14]. Additionally, the longitudinal configuration of the flow channel such as zigzag, curvy, step [15], single-layer, double-layer, tapered [16], wavy [17] and converting channels [18] for the same cross-sectional shape was investigated and

compared with the conventional straight channel with uniform cross-sectional shape.

On this background, the comparison between different cross-sectional shapes to accomplish the two objectives simultaneously, lower pressure drop along the flow channel and higher heat transfer rate between the fluid flow and the heated wall, is a new point of view that deserves to be investigated. In this study we seek to decrease simultaneously the longitudinal flow resistance and the transversal thermal resistance for the square, circle and triangle cross-section channels. The work is based on two analytical models, i.e., scale analysis, and computational fluid dynamics (CFD) models.

2. Square element

We begin with a square array, with these known features: side L_s , loop cross section L_s^2 , flow cross section A_f , solid cross section $A_s = L_s^2 - A_f$, known fluid properties: (k_f, μ, ν) , and solid properties: $(k_s, \text{etc.})$. The search is for the best shape of the wetted perimeter, such that two objectives are met at the same time: small transversal thermal resistance between the fluid stream and the square perimeter, and small longitudinal flow resistance, $\Delta P / \dot{m}$. The mass flow rate $\dot{m}$ is fixed; it is the stream allocated to one element with square cross section, as shown in Fig. 1. The shape of the wetted perimeter varies between the two extremes shown in the upper part of Fig. 1: (a) square flow cross section, and (b) round flow cross section.

2.1 Flow resistance. Assume that the flow is laminar and fully developed (Poiseuille). This means that the longitudinal flow resistance is [1]

$$\frac{dP}{dx} = f \frac{4}{D_h} \frac{1}{2} \rho U^2 \quad (1)$$

where U is the average fluid velocity,

$$U = \frac{\dot{m}}{\rho A_f} \quad (2)$$

D_h is the hydraulic diameter,

$$D_h = \frac{4A_f}{p} \quad (3)$$

p is the wetted perimeter, f is the friction coefficient

$$f = \frac{Po}{Re_{D_h}}, \quad Re_{D_h} = \frac{UD_h}{\nu} \quad (4)$$

and Po is the Poiseuille constant. The extreme values of Po for cases (a) and (b) in Fig. 1 are [1], respectively:

$$Po_a = 14.2 \quad Po_b = 16 \quad (5)$$

The effect of the shape of the wetted perimeter is brought into the analysis by introducing the dimensionless factors ϕ and ψ :

$$\begin{aligned} A_f &= \phi L_s^2, & \phi &< 1 \\ p &= \psi 4L_s, & \psi &< 1 \end{aligned} \quad (6)$$

Combining Eqs. (1) – (6), we arrive at the dimensionless fluid flow resistance

$$R_f = \frac{dP/dx}{\dot{m}\nu / D_h^2 A_f} = 2Po \frac{\psi^2}{\phi^3} \quad (7)$$

One objective is to decrease R_f . We know already that if this is the only objective then the best cross section is (b), the round cross section. In this study, however, there is also a second objective (to minimize the transversal thermal resistance), and for this reason we continue with the general case where ϕ and ψ may vary.

When the rule of morphing (a) into (b) is simple, as in Fig. 1 where all the corners are identical, there is a relation between ϕ and ψ . To see this, assume that ϕ is specified, therefore $\phi_a = \phi_b = \phi$. Next, for the round perimeter (b) we have $p_b = \pi D_b$ and $A_b = (\pi/4)D_b^2$, from which we deduce

$$\psi_b = \frac{\pi^{1/2}}{2} \phi^{1/2} \quad (8)$$

In case (a), the solid cross section is the same as in case (b), but $p_a = 4L_a$ and $A_a = L_a^2$, therefore

$$\psi_a = \phi^{1/2} \quad (9)$$

Another factor that could have been varied is the ratio R/L_c (i.e., case c) while keeping the minimum wall thickness fixed. Varying the radius of curvature was not considered as both the dimensionless fluid flow resistance and thermal resistance would fall within the limits given by cases (a) and (b). To draw case (c), we assume that the radius of the rounded corners is $R = 0.1L_c$, so that $p_c = 4L_c + 2\pi R$ and $A_c = L_c^2 + 4RL_c + \pi R^2$, therefore

$$\psi_c = 0.967 \phi^{1/2} \quad (10)$$

In summary, the values of $R_{f,a}$, $R_{f,b}$ and $R_{f,c}$ depend only on the assumed value for ϕ . By varying ϕ , the fluid cross-section varies in size while the area of the square solid is kept constant. These functions are plotted in Fig. 2. Note that case (b) exists when ϕ does not exceed the value that corresponds to the circle touching the sides of the square in Fig. 1 (b):

$$\phi_{\max,b} = \frac{(\pi/4)L_s^2}{L_s^2} = 0.785 \quad (11)$$

Similarly, case (c) exists when ϕ does not exceed 0.994.

2.2 Thermal resistance. The thermal objective of the flow system can be simulated numerically for the model shown in Fig. 3. The system is a sufficiently long parallelepiped with a square cross section in which we can fit any of the drawings shown in Fig. 1. Each such drawing is characterized by its ϕ and ψ .

The fluid flow is laminar and fully developed, which means that the longitudinal velocity distribution (u) does not vary from one x cut to another. The temperature field is also

fully developed, therefore the meniscus shape of the T distribution over the flow cross section does not vary from one x to another.

Assume that the long external walls of the parallelepiped are isothermal at T_w , and that the average temperature of the fluid entering through the $x = 0$ end is T_{in} . Also assume that the mass flow rate ($\dot{m}$) along the duct is specified. The continuity of heat flux at the solid-fluid interface (in the same plane as the cross section) requires

$$k_s \frac{\partial T}{\partial n} = k_f \frac{\partial T}{\partial n} \quad (12)$$

where n is the normal to the solid/fluid interface. Equation (12) reveals the role of $\tilde{k} = k_s / k_f$ as a dimensionless parameter of the flow system. Another parameter is the fluid Prandtl number, Pr , set equal to 1, which is the order of magnitude of common fluids.

The overall thermal contact is measured between the outer surface of the system (T_w) and the bulk temperature of the fluid in the channel. At the outlet, $x = L_x$, the bulk fluid temperature is T_{out} . The fluid temperature change from inlet to outlet is given by (Fig. 4) [1]

$$\ln \frac{T_w - T_{in}}{T_w - T_{out}} = \frac{\bar{h} \psi 4L_s L_x}{\dot{m} c_p} \quad (13)$$

where $4L_s$ is the perimeter of the square cross section, and $\bar{h}$ is an effective (overall) heat transfer coefficient that accounts for the thermal contact between T_w and the bulk temperature $T(x)$. Note that $T(x)$ increases from T_{in} to T_{out} . The right side of Eq. (13) is the number of heat transfer units of the fluid and solid system, which is defined as

$$N = \frac{\psi 4L_s L_x \bar{h}}{\dot{m} c_p} \quad (14)$$

where $\psi 4L_s$ is the perimeter of contact, and N is greater when the thermal contact is stronger.

The non-isothermal flow COMSOL model was used to simulate the temperature field of

the channel flow. An non-isothermal model is used for the fluid flow and heat transfer, i.e., in which the fluid properties such as density and viscosity are temperature dependent. The fluid properties at the duct entrance are $c_p = 4192 \text{ J/(kgK)}$, $\rho = 999.7 \text{ kg/m}^3$, $\nu = 10^{-6} \text{ m}^2/\text{s}$. It is considered that $\dot{m}_0 = 0.01 \text{ kg/s}$. The solid wall properties are $c = 475 \text{ J/(kgK)}$, $\rho = 7850 \text{ kg/m}^3$, $k_s = 44.5 \text{ W/(mK)}$. The flow regime is laminar as the Reynolds number ($Re = UD_h/\nu$) does not exceed 2300. Here, D_h is either the diameter of the cylinder or the hydraulic diameter of the fluid flow channel.

The governing equations for the non-isothermal 3-D flow are mass flow continuity, momentum and energy equations for the fluid in the duct plus the heat conduction equation for the solid wall [19]. The boundary conditions are as follows. At the inlet of the duct, a uniform temperature T_{in} (300 K) is assumed and the mean velocity of the channel flow is specified as $U = \dot{m} / \rho A_f$. The entrance region for laminar pipe flow is calculated as $L_e / D_h \cong 0.06 Re$, where D_h is the hydraulic diameter. At the duct outlet, the pressure is constant and the heat conduction condition is $\mathbf{n} \cdot (k_f \nabla T_f) = 0$. The no slip boundary condition ($u = 0$) is applied on the side walls of the flow duct. A constant temperature boundary condition (350 K) is specified at the outer surfaces of the solid walls. Table 1 shows the computational results for one case: $\phi = 0.7$, in square arrays. The results show that the average temperature at the channel outlet becomes almost independent of the mesh size. Therefore, the mesh with 2×10^5 elements is used in this study because the computational results converge as the mesh is refined, i.e., the effect of the mesh on the computational results becomes negligible as the grid size is decreased beyond the 2×10^5 elements.

Table 1. Mesh accuracy test.

Number of elements	$T_{out,avg}$ (K)	Difference (%)
51,924	325.7256	-----
143,936	325.1887	0.165
204,771	325.1928	0.0013
311,025	325.1934	0.00018
577,148	325.1938	0.00012

In this numerical work, N is a measure of the degree of thermal contact, because it is proportional to the thermal conductance, cf. Eq. (14). The value of N is determined after numerical simulations, i.e., after calculating T_{out} and substituting this value (along with T_w and T_{in}) on the left side of Eq. (13). There will be one $N_a(\phi)$ curve for the configuration of Fig. 1a, one $N_b(\phi)$ for Fig. 1b and another $N_c(\phi)$ for Fig. 1c. Keeping all the other parameters fixed, such as $\dot{m}$, Pr , $\tilde{k}$ and c_p , and varying only ϕ , it is expected that N would decrease when the fluid flow dimensionless cross-section (ϕ) would increase.

Qualitatively, the overall thermal resistance R_{th} varies monotonically with the inverse of N , as reported in Fig. 5,

$$R_{th} = \frac{1}{N} \quad (15)$$

The trend in Fig. 5 is the opposite of what it was anticipated for R_f in Fig. 2. This conflict is the basis for the two-objective search sketch in Fig. 6, which is obtained by eliminating ϕ between Figs. 2 and 5. The search for better configurations [(a), (b), or (c)] is toward low- R_f and low- R_{th} simultaneously. The better configuration is the one represented by a curve that comes closer to the origin, namely, the curve for the cross-section shown in Fig. 1a.

3. Triangular element

The analysis of the two-objective performance (R_f , R_{th}) of a triangular array follows the same steps as in the preceding section. We focus on the extreme configurations, (a) triangular flow channel and (b) round flow channel, while the volume of wall material is fixed. The volume fraction occupied by the flow cross-section and channel is ϕ . The flow cross-section and the wetted perimeter are $A_t = 0.433\phi L_t^2$ and $p = \psi 3L_t$, such that $\psi_a = \phi^{1/2}$ and $\psi_b = 3^{-3/4} \pi^{1/2} \phi^{1/2}$.

3.1 Flow resistance. In place of Eq. (7) we obtain the following dimensionless fluid flow resistance

$$R_f = \frac{dP / dx}{\dot{m} v / D_h^2 A_f} = \frac{9}{8} Po \frac{\psi^2}{\phi^3} \quad (16)$$

The mass flow rate $\dot{m}$ is fixed: In this case, $\dot{m}$ is the stream allocated to one element with triangular cross section, Fig. 7. For the triangular duct, Fig. 7a, we substitute $Po_a = 13.3$ and conclude that

$$R_{f,a} = 14.96 \phi^{-2} \quad (17)$$

Similarly, for the round cross section, Fig. 7b, where $Po_b = 16$, we obtain

$$R_{f,b} = 10.88 \phi^{-2} \quad (18)$$

Equations (17) and (18) are plotted in Fig. 8, which shows that the flow resistance of the round channel is smaller than the resistance posed by the triangular channel. Note the similarity between Fig. 8 and Fig. 2. For the round cross section (b), the $R_f(\phi)$ curve ends at $\phi = 0.6$, which represents the round duct that touches the triangular perimeter of the available space.

3.2 Thermal resistance. The effect of duct cross-sectional shape on the thermal resistance was determined by using the same numerical method as in section 2.2 and Fig. 2. The effective number of heat transfer units of the fluid-solid flow system that occupies the volume with triangular cross section and length L_x . The number of heat transfer units is based on the

perimeter of the triangular cross section of the volume,

$$N = \frac{\psi 3L_t L_x \bar{h}}{\dot{m} c_p} \quad (19)$$

The inverse of N is regarded as a measure of the degree of thermal contact between the fluid and the external perimeter ($3L_t$), namely $R_{th} = N^{-1}$, as shown in Fig. 9. The thermal resistance of the design with triangular flow cross section (a) is smaller than the thermal resistance of the design with round flow cross section.

Figures 8 and 9 were combined into the $R_f - R_{th}$ plot shown in Fig. 10. For designs with two simultaneous objectives, small R_f and small R_{th} , configuration (a) is better than configuration (b). This conclusion corresponds with what it was observed in Fig. 6 for square cross sections.

4. Square array or triangular array?

The similarities between the performance of the square elemental volume (Figs. 2, 5, 6) and the triangular elemental volume (Figs. 8-10) are evident. They invite the question of which cross-section would performs better, the square or equilateral triangle.

To answer this, first we recognize that the whole assembly is an array of many elements of the same kind (Fig. 11) in which the two fluid elements are arranged in a staggered manner and the heat exchanger is of counterflow type. The whole assembly has a purpose and a flow rate, which are the same regardless of the choice made for the shape of the elemental cross-section. In other words, the mass flow rate per unit of *array* cross sectional area, $\dot{m}''$ is specified (fixed).

Second, in the study of square elements (Fig. 1), $\dot{m}$ was fixed because the size (the cross section) of the square element was fixed. The comparison was between three flow cross sectional shapes (a, b, c), each carrying the same $\dot{m}$. This is why the flow resistance defined in Eq. (7) depended on only ϕ , ψ and Po . The same is true for the flow resistance defined in Eq.

(16) for the triangular element: the comparison was between Figs. 7a and 7b, and $\dot{m}$ was the same for both configurations.

As we move to the whole arrays shown in Fig. 11, we note not only that $\dot{m}''$ is the same for the two arrays, but also that the $\dot{m}$ that appears in Eq. (7) is not necessarily the same as the $\dot{m}$ in Eq. (16). For clarity, we compare the two arrays in terms of pressure drop, not flow resistances. For the square array, Eq. (7) can be rewritten as

$$\left(\frac{dP}{dx} \right)_s \frac{L_x^2}{\dot{m}'' v} = \frac{2 Po_s}{\phi^4 \tilde{L}_s^2} \quad (20)$$

where $\tilde{L}_s = L_s / L_x$ is the dimensionless side of the square element, and L_x is the length of the array in the flow direction.

For the triangular array, Eq. (16) can be rewritten as

$$\left(\frac{dP}{dx} \right)_t \frac{L_x^2}{\dot{m}'' v} = \frac{27}{8} \frac{Po_t}{\phi^4 \tilde{L}_t^2} \quad (21)$$

where $\tilde{L}_t = L_t / L_x$ is the dimensionless side of the triangular element. Dividing Eqs. (20) and (21), and recalling $Po_s = 14.2$ and $Po_t = 13.3$, we obtain

$$\frac{(dP/dx)_s}{(dP/dx)_t} = 0.633 \left(\frac{L_t}{L_s} \right)^2 \quad (22)$$

The ratio L_t/L_s is a degree of freedom in the drawing shown in Fig. 11:

(i) If the sides of the elements are equal, $L_t = L_s$, then the pressure drop along the square array is 36.7 percent smaller than along the triangular array.

(ii) If the cross sectional areas of the elements are equal, $L_s^2 = (3^{1/2} / 4) L_t^2$, then the right side of Eq. (22) is equal to 1.461. In this case, the pressure drop along the triangular array is one third smaller than along the square array.

The thermal resistances of the two arrangements based on square and triangular cross-sections can also be compared on an equal basis. To start with, as noted already en route to Eq. (15), the thermal resistance of one element is large when N is small. The relation between thermal resistance and $1/N$ becomes a proportionality in the limit $N \ll 1$. This can be shown from Eq. (13), after writing N for the right hand side, and $\Delta T = T_w - T_{in}$, and also $q = \dot{m}c_p(T_{out} - T_{in})$, where q is the total heat transfer rate from the element of length L_x to the stream $\dot{m}$ allocated to that element. In the limit $N \ll 1$, Eq. (13) becomes

$$\frac{\Delta T}{q / \dot{m}c_p} = \frac{1}{N} \quad (23)$$

and, after returning to the notation of Eq. (13),

$$\left(\frac{\Delta T}{q} \right)_s = \frac{1}{\bar{h}_s \psi 4 L_s L_x} \quad (24)$$

The corresponding formula for the triangular element, derived from Eq. (19), is

$$\left(\frac{\Delta T}{q} \right)_t = \frac{1}{\bar{h}_t \psi 3 L_t L_x} \quad (25)$$

Dividing Eq. (24) and (25), we arrive at the ratio of the two thermal resistances,

$$\frac{(\Delta T/q)_s}{(\Delta T/q)_t} = \frac{3}{4} \frac{\bar{h}_t}{\bar{h}_s} \frac{L_t}{L_s} \quad (26)$$

Next, for fully developed laminar flow in channels with constant temperature [as in Eq. (13)], we have from Ref. [1], Table 3.2,

$$Nu_s = \frac{\bar{h}_s D_{h,s}}{k_f} \cong 2.89 \quad Nu_t = \frac{\bar{h}_t D_{h,t}}{k_f} \cong 2.35 \quad (27)$$

where $D_{h,s} = \psi L_s$ and $D_{h,t} = \psi 3^{-1/2} L_t$, cf. Table 3.1 in Ref. [1].

The right side of Eq. (26) turns out to be equal to 1.056, independently of the choice of L_t/L_s . In conclusion, from a thermal contact standpoint, the triangular array (Fig. 7a) is 5.3 percent superior to the square array (Fig. 1a).

5. Discussion and Conclusions

In this article we illustrated the search for optimum cross-section of the flow in an assembly with identical array of tubes in counterflow with two objectives at the same time: reduced fluid flow resistance and reduced heat transfer resistance. We performed the scale analysis for individual elements (pipe flow). The fluid flow and heat transfer was simulated using a 2D model. The element cross sections considered in this study were square, circular, and equilateral triangular. The element wall material was fixed, while the shape and length of the wetted perimeter of the flow channel could vary.

For the square elements, we found that the geometry that serves the two objectives best is the one with constant wall thickness, i.e., square wetted perimeter (Fig. 1a). The design with circular wetted perimeter (Fig. 1b) is the worst. The designs with wetted perimeters shaped as squares with rounded corners are in between. It was also found that the triangular wetted perimeter cross-section (Fig. 7a) outperforms the one with circular flow cross section (Fig. 7b).

Important and interesting is the comparison between the two arrays consisting of square and triangular elements (Fig. 11). The comparison is done on the basis of a specified fluid mass flow rate per unit of array cross sectional area, such that the ratio L_s/L_t is a degree of freedom. We considered two such ratios, $L_s/L_t = 1$, and $L_s/L_t = 3^{1/4}/2$. In summary, the array of triangular channels meets the two objectives better than the array of square channels.

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Figure captions

- Figure 1 Three designs of flow channels embedded in a square cross section.
- Figure 2 The variation of fluid flow resistance for different flow channels, represented by area factor ϕ , of square arrays, (a) square and (b) circular flow channel.
- Figure 3 Velocity and temperature profiles of a flow channel embedded in square solid.
- Figure 4 The mean temperature of the fluid stream which flows through hot contacting walls.
- Figure 5 The variation of overall thermal resistance with area factor ϕ , (a) square flow channel and (b) circular flow channel.
- Figure 6 The overall thermal resistance versus the fluid flow resistance of (a) square flow channels, (b) circle flow channels.
- Figure 7 Flow channels embedded in a triangular cross section.
- Figure 8 The flow resistance variation for triangular arrays with the area factor ϕ , (a) triangle flow channel and (b) circular flow channel.
- Figure 9 For triangular cross-section; the overall thermal resistance versus area factor ϕ .
- Figure 10 The results of fluid flow resistance and thermal resistance for the triangular and circular flow channels embedded in triangle cross section.
- Figure 11 Square and triangle arrays with the same flow length and mass flow rate per unit of cross-section area.

Figure 1

(a)

(b)

Figure 2

Figure 3

Figure 4

Figure 5

Figure 6

Figure 7

(a)

(b)

Figure 8

Figure 9

Figure 10

Figure 11

Graphical Abstract